

Note:Cyclostationary Verification

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A. Radiated sound from a moving body

The FW-H equation is now a widely used starting point for theories of noise generation by moving bodies like propellers in a moving medium, even when turbulence noise is of little or no importance, given as [1]

$$\left[\frac{1}{c_0^2} \frac{D^2}{Dt^2} - \nabla^2 \right] \{p'(\vec{x}, t)H(f)\} = \underbrace{\frac{\partial}{\partial x_i \partial x_j} [T_{ij}H(f)]}_{\text{quadrupole}} - \underbrace{\frac{\partial}{\partial x_i} L_i \delta(f)}_{\text{loading}} + \underbrace{\frac{D}{Dt} [Q\delta(f)]}_{\text{thickness}} \quad (1)$$

with

$$\begin{aligned} T_{ij} &= \rho u_i u_j + [(p - p_0) - c_0^2(\rho - \rho_0)]\delta_{ij} - \sigma_{ij} \\ L_i &= \rho u_i [u_n - (v_n - U_{\infty n})] + P_{ij} \hat{n}_j \\ Q &= \rho [u_n - (v_n - U_{\infty n})] + \rho_0 (v_n - U_{\infty n}) \end{aligned} \quad (2)$$

$H(f)$ denotes the Heaviside function, which is used to vanish identically within the moving body.

The convective derivative $\frac{D}{Dt}$ is defined by

$$\frac{D}{Dt} = \frac{\partial}{\partial t} + \vec{U}_{\infty} \cdot \nabla \quad (3)$$

where \vec{U}_{∞} is the vector of mean flow velocity, and ∇ the divergence operator.

Combined with the general form the Green's function with impulsive point sources

$$\left[\frac{1}{c_0^2} \frac{D^2}{Dt^2} - \nabla^2 \right] G(\vec{x}, t; \vec{y}, \tau) = \delta(\vec{x} - \vec{y})\delta(t - \tau) \quad (4)$$

Here we consider the solution of Green's function with arbitrary orientation of \vec{U}_{∞} as extend the case of propeller noise to an angle of attack [2].

$$G(\vec{x}, t; \vec{y}, \tau) = \frac{\delta(\tau - t + R/c_0)}{4\pi R^*} \quad (5)$$

The quantities R^* and R are defined as

$$R^* = \frac{r}{\gamma} \sqrt{1 + \gamma^2 M_{\infty r}^2} \quad (6)$$

and

$$R = \gamma^2 (R^* - r M_{\infty r}^2) \quad (7)$$

where $\gamma^2 = 1/(1 - M_{\infty}^2)$ and $M_{\infty r} = \vec{M}_{\infty} \cdot \hat{r}$ with \hat{r} being the unit vector of the distance between the source position $y(\tau)$ and the observer position $x(t)$, i.e. $r = |x(t) - y(\tau)|$.

Thus, substitute the right hand side of Eq.(1) as the source terms, the solution of FW-H equation can be derived with the property of Green's function

$$\begin{aligned} H(f)p'(\vec{x}, t) &= \int_{-\infty}^t \int_v q(\vec{y}, \tau) G(\vec{x}, t; \vec{y}, \tau) d^3 \vec{y} d\tau \\ &= \int_{-\infty}^t \int_v \left\{ \frac{\partial}{\partial x_i \partial x_j} [T_{ij} H(f)] - \frac{\partial}{\partial x_i} L_i \delta(f) + \frac{D}{Dt} [Q \delta(f)] \right\} \frac{\delta(\tau - t + R/c_0)}{4\pi R^*} d^3 \vec{y} d\tau \end{aligned} \quad (8)$$

B. modeling of propeller noise

For modeling propeller noise, we only concentrate on the loading (2nd term) and thickness (3rd term) contributions, the nonlinear term T_{ij} is ignored. The Eq.(8) is simplified as follows:

$$p'(\vec{x}, t) = \int_{-\infty}^t \int_S \left\{ -\frac{\partial}{\partial x_i} L_i \delta(f) + \frac{D}{Dt} [Q \delta(f)] \right\} \frac{\delta(\tau - t + R/c_0)}{4\pi R^*} dS d\tau \quad (9)$$

In order to intergrate over $d\tau$, the method of generalized function theory of the Farassat is used:

$$\int_{-\infty}^t h(\tau) \delta(g) = \left[\frac{h(\tau)}{|\partial g / \partial \tau|} \right]_e \quad (10)$$

where $[\cdot]_e$ is the notion denotes evaluation of the term with $g = 0$, $t = \tau + R/c_0$ as regard time.

The partial differentiation of g with respect to τ yields

$$\frac{\partial g}{\partial \tau} = 1 + \frac{1}{c_0} \frac{\partial R}{\partial \tau} = 1 - M_i \frac{\partial R}{\partial x_i} \quad (11)$$

Here, we only give the final result used for cyclostationarity modeling. More complete derivation can be found in Ghorbaniasl's paper[2].

$$\begin{aligned}
4\pi p'_L(\vec{x}, t) &= \frac{1}{c_0} \frac{\partial}{\partial t} \int_S \left[\frac{L_R}{R^*(1 - M_R)} \right]_e dS + \int_S \left[\frac{L_{R^*}}{R^{*2}(1 - M_R)} \right]_e dS \\
4\pi p'_T(\vec{x}, t) &= \frac{\partial}{\partial t} \int_S \left[\frac{(1 - M_{\infty R})Q}{R^*(1 - M_R)} \right]_e dS - \int_S \left[\frac{c_0 M_{\infty R^*} Q}{R^{*2}(1 - M_R)} \right]_e dS
\end{aligned} \tag{12}$$

where $L_R = L_i \frac{\partial R}{\partial x_i}$, $L_{R^*} = L_i \frac{\partial R^*}{\partial x_i}$, $M_{\infty R} = M_{\infty i} \frac{\partial R}{\partial x_i}$, $M_{\infty R^*} = M_{\infty i} \frac{\partial R^*}{\partial x_i}$, $M_R = M_i \frac{\partial R}{\partial x_i}$.

The formulation of loading noise p'_L and thickness noise p'_T take account the convection effect and incidence of mean flow, and is equivalent to the Farassat's formulation 1[3] if the medium is at rest, $\vec{M}_\infty = 0$. The term of $(1 - M_R)^{-1}$ refers as the Doppler factor.

References

- [1] Ffowcs Williams JE, Hawkings DL "Sound generation by turbulence and surfaces in arbitrary motion". Phil. Trans. R. Soc. Lond. A, **264**, 1969, 321-342.(doi:10.1098/rsta.1969.0031)
- [2] Ghorbaniasl G, Lacor C "A moving medium formulation for prediction of propeller noise at incidence". J. Sound Vib, **331**, 2012, 117-137.(doi:10.1016/j.jsv.2011.08.018)
- [3] Farassat F, Myers MK "Multidimensional generalized functions in aeroacoustics and fluid mechanics" - part 1: basic concepts and operations". Int. J. Aeroacoust, **10**(1), 2011, 161-200.(doi:10.1260/1475-472X.10.2-3.161)